## **Problem description**

In this problem, we analyze a simple beam structure. Each part of this lesson shows the solution of one of the analyses. In part 1, we demonstrate the following topics:

- Starting up/shutting down the AUI
- Defining geometry points
- Defining a geometry line
- Applying a boundary condition
- Defining a material
- Defining a cross-section
- Defining and applying a load
- Generating an element
- Generating an ADINA Structures data file
- Saving an ADINA-IN database file
- Running ADINA Structures
- Loading an ADINA Structures porthole file
- Drawing a mesh plot, with boundary conditions plotted
- Drawing a load plot
- Listing the extreme deflections
- Plotting bending moment and shear force diagrams

In part 2, we demonstrate the following additional topics:

- Opening an ADINA-IN database file to access a previously defined model
- Deleting and redefining a load

In part 3, we demonstrate the following additional topics:

- Adding a boundary condition to an existing model
- Deleting elements from the model
- Clicking on a green column to fill in a table.
- Subdividing a line, generating elements on a subdivided line
- Magnifying the plotted displacements
- Listing the nodal point forces and moments

In part 4, we demonstrate the following additional topics:

- Obtaining internal section forces and moments
- Listing the internal section point forces and moments

In part 5, we demonstrate the following additional topics:

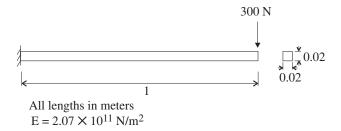
• Using the fixed-end force correction feature

## Before you begin

Please refer to the Icon Locator Tables chapter of the Primer for the locations of all of the AUI icons. Please refer to the Hints chapter of the Primer for useful hints.

This problem can be solved with the 900 nodes version of the ADINA System.

## Part 1: Deflection due to a tip load



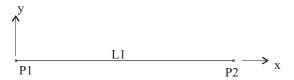
#### Invoking the AUI and choosing the finite element program

Invoke the AUI and set the Program Module drop-down list to ADINA Structures.

(See the hints in the Hints chapter for more information.)

#### **Defining model geometry**

When setting up a model, it is frequently useful to make a diagram showing the key geometry used in the model definition, such as the following diagram:



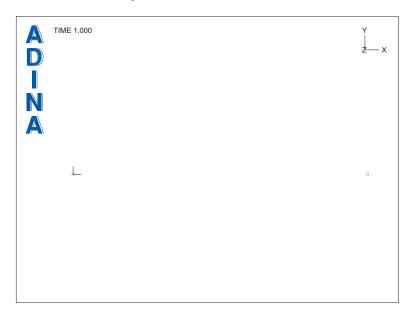
You may find it useful to refer to this diagram when working through this problem.

Click the Define Points icon And enter the following information into the table:

Point #	X1	X2	X3
1			
2	1		

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(Table cells that are left blank are treated as zeros.) Then click OK. The graphics window should look something like this:



Now click the Define Lines icon and click the Add... button to add line number 1. Set the Point 1 field to 1, the Point 2 field to 2 and click OK. The graphics window should look something like the top figure on the next page.

#### **Defining boundary conditions**

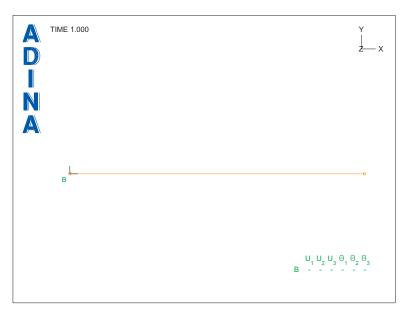
Click the Apply Fixity icon , enter 1 in the first row of the Point # table and click OK.

Click the Boundary Plot icon . The graphics window should look something like the bottom figure on the next page.

#### **Defining loads**

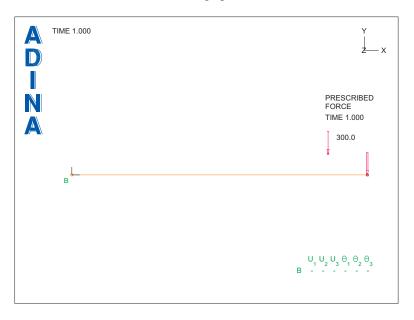
Click the Apply Load icon to open the Apply Load dialog box. Make sure that the Load Type is Force and click the Define... button to the right of the Load Number field. In the Define Concentrated Force dialog box, add Concentrated Force Number 1, set the Magnitude to 300, the Y Force Direction to -1 and click OK. In the first row of the table in the Apply Load dialog box, set the Point # to 2 and click OK to close the Apply Load dialog box.





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Click the Load Plot icon . The graphics window should look something like this:



## **Defining the cross-section**

Click the Cross Sections icon . Then add section number 1, set the Width to 0.02, click the Square Section button and click OK.

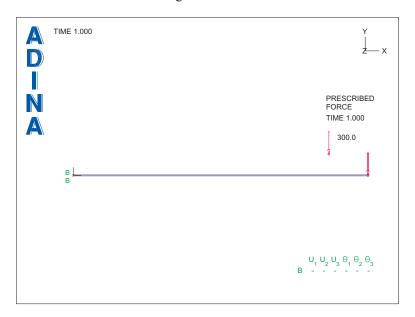
### **Defining the material**

Click the Manage Materials icon and click the Elastic Isotropic button. In the Define Isotropic Linear Elastic Material dialog box, add material 1, set the Young's Modulus to 2.07E11 and click OK. Click Close to close the Manage Material Definitions dialog box.

## **Defining the finite elements**

*Element group:* Click the Element Groups icon , add group 1, set the Type to Beam and click OK.

Element generation: Click the Mesh Lines icon , set the components of the Orientation Vector to (0, 1, 0), enter 1 in the first row of the Line # table and click OK. (The orientation vector is used to define the directions of the element local coordinate system; the element s-direction lies in the plane defined by the element and the orientation vector.) The graphics window should look something like this:



# Generating the ADINA Structures data file, running ADINA Structures, loading the porthole file

First click the Save icon and save the database to file prob01 (the "File type" field should be "ADINA-IN Database Files (\*.idb)"). To generate the ADINA Structures data file and run

ADINA Structures, click the Data File/Solution icon , set the file name to prob01, make sure that the Run Solution button is checked and click Save.

When ADINA Structures is finished, it displays the message

"\* Solution successful, please check the results \*"

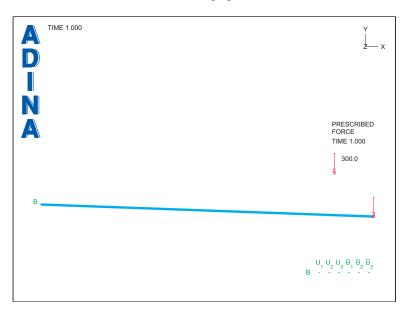
in the ADINA Structures Message Window. Close all open dialog boxes.

Set the Program Module drop-down list to Post-Processing, click Yes to discard all changes and continue, click the Open icon and open porthole file prob01.

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## Displaying the deformed model

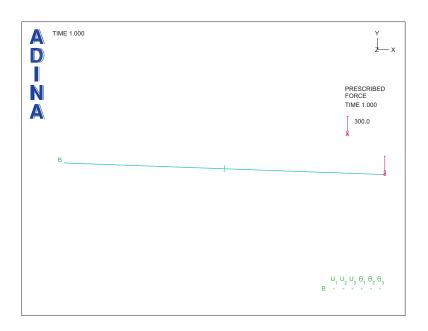
Click the Boundary Plot icon to display the boundary conditions. Then click the Load Plot icon to show the load. The graphics window should look something like this:

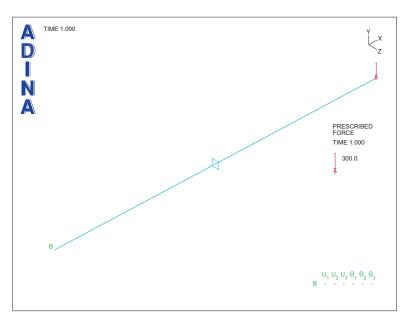


The beam cross-section can be plotted at the center of the element. Click the Modify Mesh Plot icon (or right-click on the mesh plot and choose Modify Mesh Plot ), click the Element Depiction... button, click the Display Beam Cross-Section field, then click OK twice to close both dialog boxes. The graphics window should look something like the top figure on the next page.

Click the Iso View 2 icon (located in the drop-down list next to the YZ View icon to plot the beam from a different angle. The graphics window should look something like the bottom figure on the next page.

Click the XY View icon to return to the original view.





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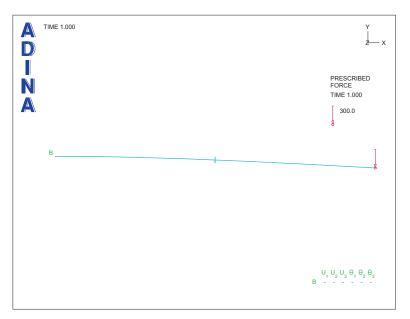
#### Listing the tip deflection

Choose List—Extreme Values—Zone. In the "Variables to List" box, first row, choose Y-DISPLACEMENT from the right-hand drop-down list (the list with the downwards pointing arrow). Then click Apply.

The AUI displays the value -3.62319E-02 for the y-displacement of node 2. Note that this is the deflection predicted by beam theory; a single beam element is sufficient in this case because the beam element contains a cubic displacement assumption and the beam theory solution requires a cubic displacement assumption. Click Close to close the dialog box.

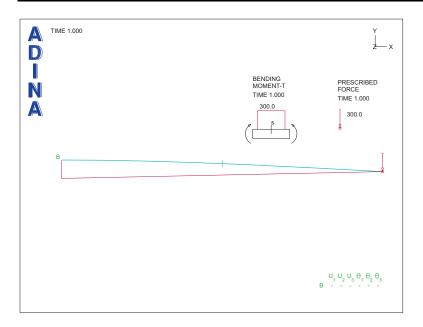
In order to see the cubic displacements of the neutral axis, click the Modify Mesh Plot icon

Neutral Axis' to 8, then click OK twice to close both dialog boxes. The graphics window should look something like this:

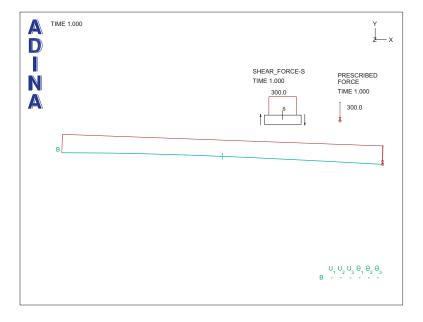


## Plotting the bending moment diagram and the shear force diagram

Bending moment diagram: Choose Display—Element Line Plot—Create, set the Element Line Quantity to BENDING\_MOMENT-T and click OK. The graphics window should look something like the top figure on the next page.



Shear force diagram: Choose Display—Element Line Plot—Modify, set the Element Line Quantity to SHEAR\_FORCE-S and click OK. The graphics window should look something like this:



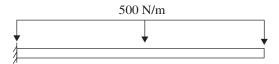
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#### **Exiting the AUI**

Choose File→Exit and click Yes to discard all changes and exit ADINA-AUI.

#### Part 2: Deflection due to a distributed load

Now we apply a distributed load to the same cantilever as shown:



Invoke the AUI and set the Program Module drop-down list to ADINA Structures. Then choose file prob01.idb from the recent file list near the bottom of the File menu (File→Recently Opened Files for the Linux version).

#### Deleting and redefining the load

*Deleting the load:* In the Model Tree, click on the + next to the Loading text, right-click on the "1. Force 1 on Point 2" text, choose Delete, and click Yes to answer the prompt.

Click the Redraw icon to update the graphics window. The graphics window should look something like the top figure on the next page.

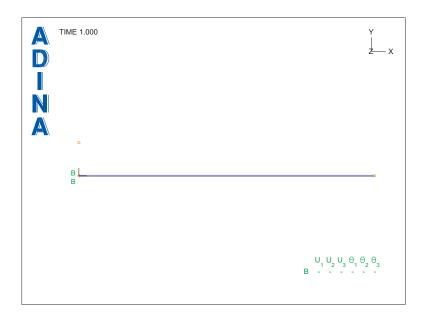
Auxiliary point: In order to define the distributed load, we need to define an auxiliary point.

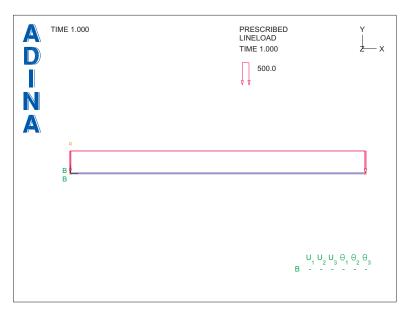
Click the Define Points icon , add point 3 and click OK:

Point #	X1	X2	X3
3		0.1	

Redefining the load: Click the Apply Load icon and set the Load Type to Distributed Line Load. Click the Define... button to the right of the Load Number field. In the Define Distributed Line Load dialog box, add line load number 1, set the Magnitude [Force/Length] to 500 and click OK. In the first row of the table in the Apply Load dialog box, set the Line # to 1 and the Aux. Point to 3, then click OK.

When you click the Redraw icon , the graphics window should look something like the bottom figure on the next page.





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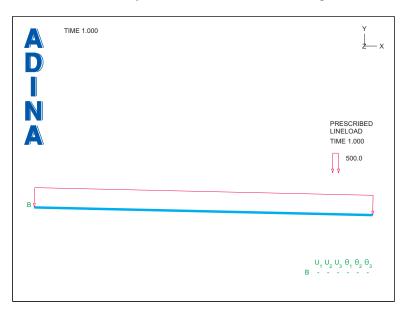
#### Generating the ADINA Structures data file, running ADINA Structures, post-processing

To save the ADINA-IN database to a new file, choose File→Save As, set the file name to prob01a and click Save.

Click the Data File/Solution icon , set the file name to prob01a, make sure that the Run Solution button is checked and click Save.

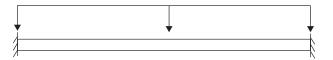
When ADINA Structures is finished, close all open dialog boxes.

Post-process this model using exactly the same steps as with the previous model (except that you will load porthole file prob01a this time). This time the tip displacement is -2.26449E-02. Again this is the same tip displacement as predicted by beam theory. The deformed mesh with loads and boundary conditions should look something like this:



Part 3: Deflection of a beam built-in at both ends

Now we change the problem to that of a beam built-in at both ends subjected to a distributed load as shown:



Set the Program Module drop-down list to ADINA Structures and click Yes to discard all changes and continue. Choose prob01a.idb from the recent file list near the bottom of the File menu.

## Adding a boundary condition, refining the mesh

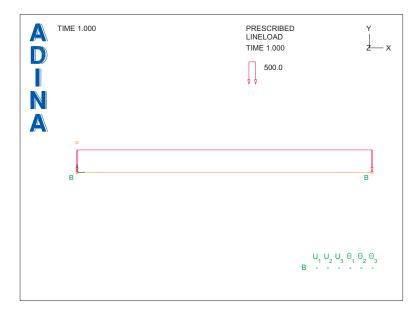
We need to add a boundary condition to the right end of the beam. In the Model Tree, highlight and right-click the Fixity text, then choose Apply.... In the Apply Fixity dialog

box, add point 2 to the second row of the table and click OK. Click the Redraw icon update the graphics.

Now we need to use more elements to solve the model.

Deleting the existing element: Click the Delete Mesh/Elements icon , enter 1 in the first row of the Line # table and click OK.

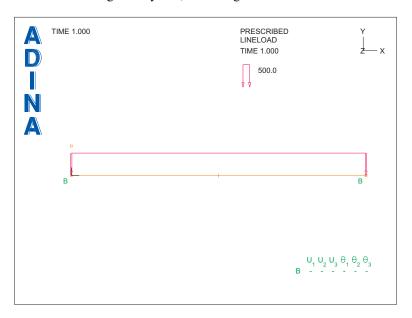
The graphics window should look something like this:



Specifying the mesh refinement: Click the Subdivide Lines icon , make sure that the "Method" is set to "Use Number of Divisions", set the Number of Subdivisions to 2 and click OK.

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The graphics window should look something like this. A short vertical line has been added to the middle of the geometry line, indicating how the line will be subdivided into elements.



Adding the elements: Click the Mesh Lines icon Vector to (0, 1, 0), enter 1 in the first row of the Line # table and click OK. (You can select the Line # using the mouse and the P button.) The graphics window should look something like the top figure on the next page.

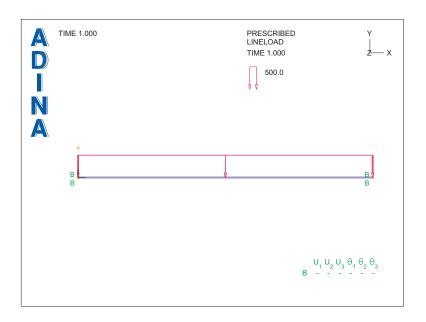
#### Generating the ADINA Structures data file, running ADINA Structures, post-processing

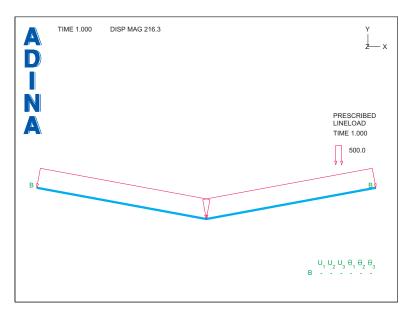
Save the database to file prob01b.idb, create ADINA Structures data file prob01b.dat and run ADINA Structures using file prob01b.dat using the same steps as with the previous models.

Post-process this model using exactly the same steps as with the previous models (except that you will load porthole file prob01b this time. This time, when you plot the deformed mesh, you will not be able to see the deformed shape because the displacements are too small.

Because the displacements are so small, we need to magnify them so that they are visible on

the plot. Click the Scale Displacements icon 107. The graphics window should look something like the bottom figure on the next page. (The plot can be improved by increasing the number of segments used to plot the neutral axis, as we demonstrated above.)

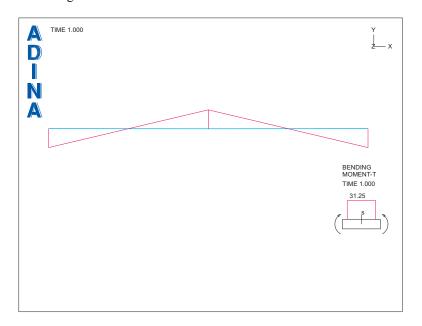




When you make the listing, note that the maximum displacement is -4.71769E-4 at the center of the model. Again this is the result predicted by beam theory.

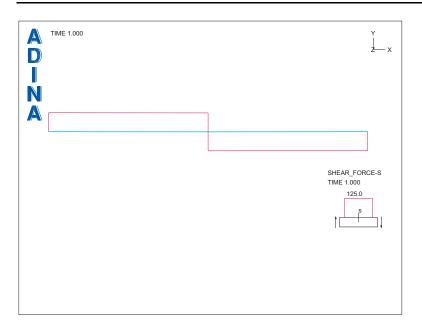
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Let's check the bending moments and shear forces. Click the Load Plot icon to remove the load plot, click the Scale Displacements icon to reduce the displacement magnification factor, then choose Display—Element Line Plot—Create, set the Element Line Quantity to BENDING\_MOMENT-T and click OK. The graphics window should look something like this:



Now choose Display—Element Line Plot—Modify, set the Element Line Quantity to SHEAR\_FORCE-S and click OK. The graphics window should look something like the figure on the next page.

These results cannot be the results predicted by beam theory, because the shear force cannot be constant along the beam when there is a distributed load acting on the beam. And the sudden change in direction of the shear force at the center of the beam is unrealistic.



In order to list the bending moments and section forces, choose List—Value List—Zone, set Variable 1 to (Force: NODAL\_MOMENT-T) (not BENDING\_MOMENT-T), set Variable 2 to (Force: NODAL\_FORCE-S) (not SHEAR\_FORCE-S) and click Apply. The listing shows

	NODAL_MOMENT-T	NODAL_FORCE-S
Element 1 of	element group 1	
Local node 1 Local node 2	3.12500E+01 3.12500E+01	1.25000E+02 -1.25000E+02
Element 2 of	element group 1	
Local node 1 Local node 2	-3.12500E+01 -3.12500E+01	-1.25000E+02 1.25000E+02

These forces are the element nodal point forces and moments, equivalent, in the virtual work sense, to the internal forces and moments. The sign convention is that given in the ADINA Structures Theory and Modeling Guide, Section 2.4, Figure 2.4-2.

In order to obtain internal forces and moments, we need to rerun the model.

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## Part 4: Computing internal forces and moments

Set the Program Module drop-down list to ADINA Structures and click Yes to discard all changes and continue. Choose prob01b.idb from the recent file list near the bottom of the File menu.

Click the Element Groups icon , set the Result Type to Section Forces, set the Number of Section Points to 5 and click OK.

#### Generating the ADINA Structures data file, running ADINA Structures, post-processing

Save the database to file prob01c.idb, create ADINA Structures data file prob01c.dat and run ADINA Structures using file prob01c.dat using the same steps as with the previous models. Then open porthole file prob01c.por.

In order to list the internal bending moments and section forces, choose List→ Value List→Zone, set Variable 1 to (Force: BENDING\_MOMENT-T), set Variable 2 to (Force: SHEAR\_FORCE-S) and click Apply. The listing shows

	BENDING MOMENT-T	SHEAR_FORCE-S
Element 1 of element 1 of element 1 of element 1	ment group 1	
Sect int pt 2 Sect int pt 3	-3.12500E+01 -1.56250E+01 0.00000E+00 1.56250E+01 3.12500E+01	-1.25000E+02 -1.25000E+02 -1.25000E+02 -1.25000E+02 -1.25000E+02
Element 2 of eleme	ment group 1	
·- · · · · · · · · · · · · · · · · · ·	3.12500E+01 1.56250E+01 0.00000E+00 -1.56250E+01 -3.12500E+01	1.25000E+02 1.25000E+02 1.25000E+02 1.25000E+02 1.25000E+02

These results agree with the nodal point forces and moments, but, again, these results cannot be correct.

## Part 5: Using fixed-end force corrections

Set the Program Module drop-down list to ADINA Structures and click Yes to discard all changes and continue. Choose prob01c.idb from the recent file list near the bottom of the File menu.

We will use the fixed-end force correction feature to improve the bending moments and shear forces. Choose Control—Miscellaneous Options, check the "Perform Fixed-End Force Corrections for Beams" button and click OK.

## Generating the ADINA Structures data file, running ADINA Structures, post-processing

Save the database to file prob01d.idb, create ADINA Structures data file prob01d.dat and run ADINA Structures using file prob01d.dat using the same steps as with the previous models. Open porthole file prob01d.por.

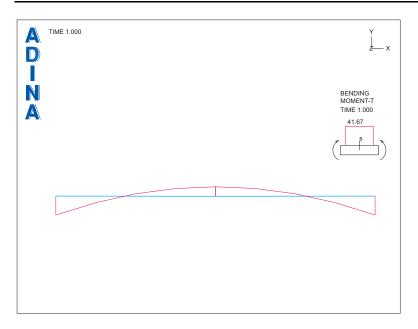
List the internal bending moments and section forces using the same steps as with the previous model. The listing shows

BENDING MOMENT-T	SHEAR_FORCE-S
ement group 1	
-4.16667E+01 -1.43229E+01 5.20833E+00 1.69271E+01 2.08333E+01	-2.50000E+02 -1.87500E+02 -1.25000E+02 -6.25000E+01 0.00000E+00
ement group 1	
2.08333E+01 1.69271E+01 5.20833E+00 -1.43229E+01 -4.16667E+01	0.00000E+00 6.25000E+01 1.25000E+02 1.87500E+02 2.50000E+02
	MOMENT-T  ement group 1  -4.16667E+01 -1.43229E+01 5.20833E+00 1.69271E+01 2.08333E+01  ement group 1  2.08333E+01 1.69271E+01 5.20833E+00 -1.43229E+01

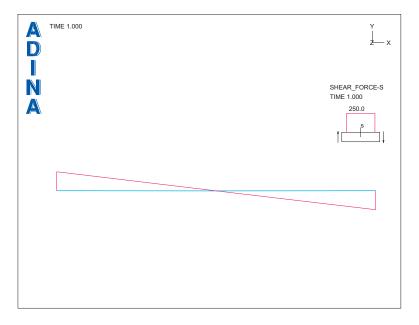
These are the results as predicted by beam theory.

Now choose Display—Element Line Plot—Create, set the Element Line Quantity to BENDING\_MOMENT-T and click OK. The graphics window should look something like the top figure on the next page.

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Now choose Display—Element Line Plot—Modify, set the Element Line Quantity to SHEAR\_FORCE-S and click OK. The graphics window should look something like this:



These are the expected results.

## **Exiting the AUI**

Choose File→Exit to exit the AUI. You can discard all changes.

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