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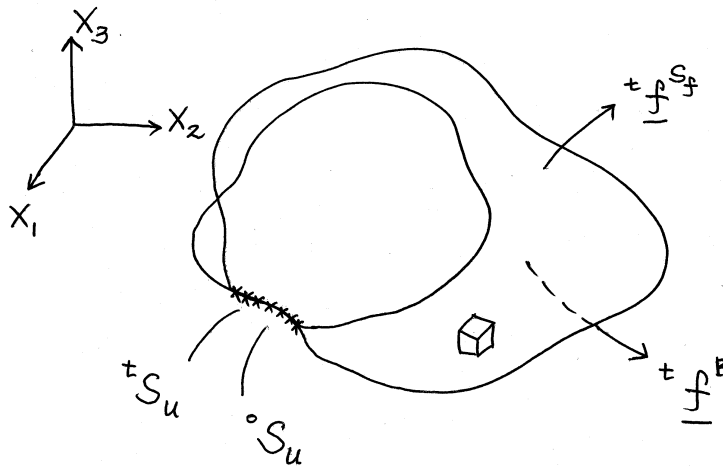
2.094 Finite Element Analysis of Solids and Fluids
Spring 2008

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Lecture 4 - Finite element formulation for solids and structures

We considered a general 3D body,

Reading:
Ch. 4



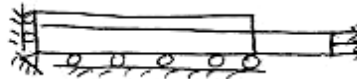
The exact solution of the mathematical model must satisfy the conditions:

- *Equilibrium* within tV and on tS_f ,
- *Compatibility*
- *Stress-strain law(s)*

I. Differential formulation

II. Variational formulation (Principle of virtual displacements) (or weak formulation)

We developed the governing F.E. equations for a sheet or bar



We obtained

$$\boxed{{}^t\mathbf{F} = {}^t\mathbf{R}} \quad (4.1)$$

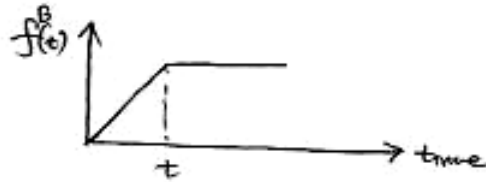
where ${}^t\mathbf{F}$ is a function of displacements/stresses/material law; and ${}^t\mathbf{R}$ is a function of time.

Assume for now linear analysis: Equilibrium within 0V and on 0S_f , linear stress-strain law and small displacements yields

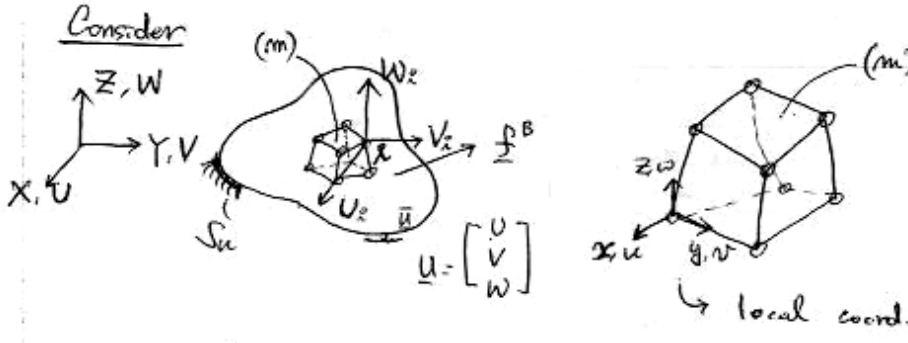
$${}^t\mathbf{F} = \mathbf{K} \cdot {}^t\mathbf{U} \quad (4.2)$$

We want to establish,

$$\mathbf{K}\mathbf{U}(t) = \mathbf{R}(t) \quad (4.3)$$



Consider



$$\hat{U}^T = [U_1 \quad V_1 \quad W_1 \quad U_2 \quad \dots \quad W_N] \quad (N \text{ nodes}) \quad (4.4)$$

where \hat{U}^T is a distinct nodal point displacement vector.

Note: for the moment “remove S_u ”

We also say

$$\hat{U}^T = [U_1 \quad U_2 \quad U_3 \quad \dots \quad U_n] \quad (n = 3N) \quad (4.5)$$

We now assume

$$\mathbf{u}^{(m)} = \mathbf{H}^{(m)} \hat{U}, \quad \mathbf{u}^{(m)} = \begin{bmatrix} u \\ v \\ w \end{bmatrix}^{(m)} \quad (4.6a)$$

where $\mathbf{H}^{(m)}$ is $3 \times n$ and \hat{U} is $n \times 1$.

$$\boldsymbol{\epsilon}^{(m)} = \mathbf{B}^{(m)} \hat{U} \quad (4.6b)$$

where $\mathbf{B}^{(m)}$ is $6 \times n$, and

$$\boldsymbol{\epsilon}^{(m)T} = [\epsilon_{xx} \quad \epsilon_{yy} \quad \epsilon_{zz} \quad \gamma_{xy} \quad \gamma_{yz} \quad \gamma_{zx}]$$

e.g. $\gamma_{xy} = \frac{\partial v}{\partial x} + \frac{\partial u}{\partial y}$

We also assume

$$\bar{\mathbf{u}}^{(m)} = \mathbf{H}^{(m)} \bar{\hat{U}} \quad (4.6c)$$

$$\bar{\boldsymbol{\epsilon}}^{(m)} = \mathbf{B}^{(m)} \bar{\hat{U}} \quad (4.6d)$$

Principle of Virtual Work:

$$\int_V \bar{\boldsymbol{\epsilon}}^T \boldsymbol{\tau} dV = \int_V \bar{\boldsymbol{U}}^T \boldsymbol{f}^B dV \quad (4.7)$$

(4.7) can be rewritten as

$$\sum_m \int_{V^{(m)}} \bar{\boldsymbol{\epsilon}}^{(m)T} \boldsymbol{\tau}^{(m)} dV^{(m)} = \sum_m \int_{V^{(m)}} \bar{\boldsymbol{U}}^{(m)T} \boldsymbol{f}^{B^{(m)}} dV^{(m)} \quad (4.8)$$

Substitute (4.6a) to (4.6d).

$$\begin{aligned} \bar{\boldsymbol{U}}^T \left\{ \sum_m \int_{V^{(m)}} \boldsymbol{B}^{(m)T} \boldsymbol{\tau}^{(m)} dV^{(m)} \right\} = \\ \bar{\boldsymbol{U}}^T \left\{ \sum_m \int_{V^{(m)}} \boldsymbol{H}^{(m)T} \boldsymbol{f}^{B^{(m)}} dV^{(m)} \right\} \end{aligned} \quad (4.9)$$

$$\boldsymbol{\tau}^{(m)} = \boldsymbol{C}^{(m)} \boldsymbol{\epsilon}^{(m)} = \boldsymbol{C}^{(m)} \boldsymbol{B}^{(m)} \hat{\boldsymbol{U}} \quad (4.10)$$

Finally,

$$\begin{aligned} \bar{\boldsymbol{U}}^T \left\{ \sum_m \int_{V^{(m)}} \boldsymbol{B}^{(m)T} \boldsymbol{C}^{(m)} \boldsymbol{B}^{(m)} dV^{(m)} \right\} \hat{\boldsymbol{U}} = \\ \bar{\boldsymbol{U}}^T \left\{ \sum_m \int_{V^{(m)}} \boldsymbol{H}^{(m)T} \boldsymbol{f}^{B^{(m)}} dV^{(m)} \right\} \end{aligned} \quad (4.11)$$

with

$$\bar{\boldsymbol{\epsilon}}^{(m)T} = \bar{\boldsymbol{U}}^T \boldsymbol{B}^{(m)T} \quad (4.12)$$

$$\boldsymbol{K} \hat{\boldsymbol{U}} = \boldsymbol{R}_B \quad (4.13)$$

where \boldsymbol{K} is $n \times n$, and \boldsymbol{R}_B is $n \times 1$.

Direct stiffness method:

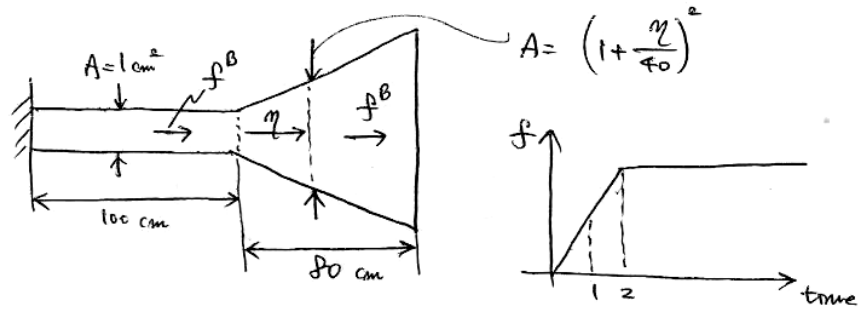
$$\boldsymbol{K} = \sum_m \boldsymbol{K}^{(m)} \quad (4.14)$$

$$\boldsymbol{R}_B = \sum_m \boldsymbol{R}_B^{(m)} \quad (4.15)$$

$$\boldsymbol{K}^{(m)} = \int_{V^{(m)}} \boldsymbol{B}^{(m)T} \boldsymbol{C}^{(m)} \boldsymbol{B}^{(m)} dV^{(m)} \quad (4.16)$$

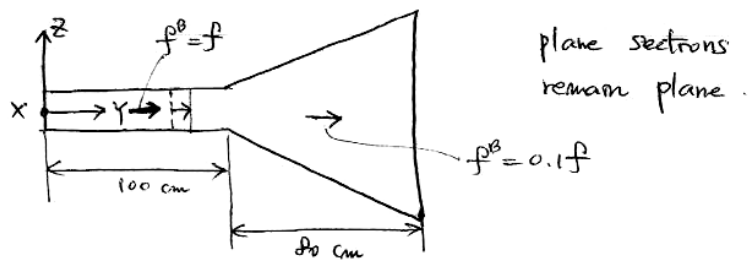
$$\boldsymbol{R}_B^{(m)} = \int_{V^{(m)}} \boldsymbol{H}^{(m)T} \boldsymbol{f}^{B^{(m)}} dV^{(m)} \quad (4.17)$$

Example 4.5 textbook

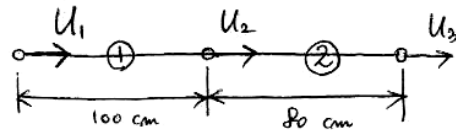


E = Young's Modulus

Mathematical model Plane sections remain plane:

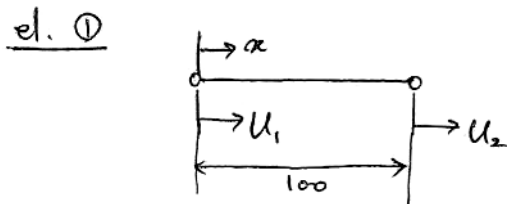


F.E. model

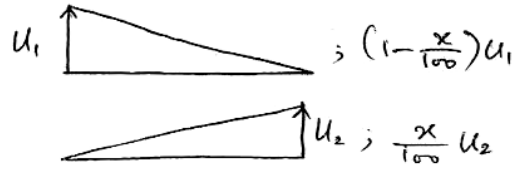


$$U = \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix} \tag{4.18}$$

Element 1



$$u^{(1)}(x) = \underbrace{\begin{bmatrix} 1 - \frac{x}{100} & \frac{x}{100} & 0 \end{bmatrix}}_{H^{(1)}} \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix} \tag{4.19}$$



$$\epsilon_{xx}^{(1)}(x) = \underbrace{\begin{bmatrix} -\frac{1}{100} & \frac{1}{100} & 0 \end{bmatrix}}_{\mathbf{B}^{(1)}} \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix} \quad (4.20)$$

Element 2

$$u^{(2)}(x) = \underbrace{\begin{bmatrix} 0 & 1 - \frac{x}{80} & \frac{x}{80} \end{bmatrix}}_{\mathbf{H}^{(2)}} \mathbf{U} \quad (4.21)$$

$$\epsilon_{xx}^{(2)}(x) = \underbrace{\begin{bmatrix} 0 & -\frac{1}{80} & \frac{1}{80} \end{bmatrix}}_{\mathbf{B}^{(2)}} \mathbf{U} \quad (4.22)$$

Then,

$$\mathbf{K} = \frac{E}{100} \begin{bmatrix} 1 & -1 & 0 \\ -1 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix} + \frac{13E}{240} \begin{bmatrix} 0 & 0 & 0 \\ 0 & 1 & -1 \\ 0 & -1 & 1 \end{bmatrix} \quad (4.23)$$

where,

$$\frac{E(1)}{100} \equiv \left(\frac{AE}{L} \right) \quad (4.24)$$

$$\frac{E \cdot 13}{3 \cdot 80} = \underbrace{\left(\frac{13}{3} \right)}_{A^*} \frac{E}{80} \quad (4.25)$$

$$A \Big|_{\eta=0} < A^* < A \Big|_{\eta=80} \quad (4.26)$$

$$1 < 4.333 < 9$$